

# Simulation of Dry Bay Discharge of Alternative Agents'

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## ABSTRACT

As a continuation of the USAF/Navy/Army/FAA sponsored halon replacement project, the discharge characteristics of the three selected alternative agents (CF<sub>3</sub>I, FC-218, and HFC-125) for dry bay applications will be further evaluated. Halon 1301 will also be included in the study for the purpose of comparison.

The experimental set-up involves a pressure vessel in which a fixed amount of agent is placed. The vessel is pressurized with nitrogen to a specified charge pressure. To simulate dry bay applications, discharge of the agent from the vessel into an unconfined space is initiated by a quick-action solenoid valve or a squib. The shape of the vessel is either cylindrical or spherical. To monitor the behavior of the agent inside the vessel during discharge, internal pressure and temperature changes are measured. A cylindrical pressure vessel with sight glasses is used for visual observations of the internal processes during discharge. The external behavior of the discharge is evaluated using a laser-light attenuation technique which has been used in the previous agent screening project and high speed photography. Parameters to be studied include: (1) nitrogen charge pressure, (2) initial amount of agent, (3) discharge temperature, (4) discharge orientation, and (5) discharge orifice *size*.

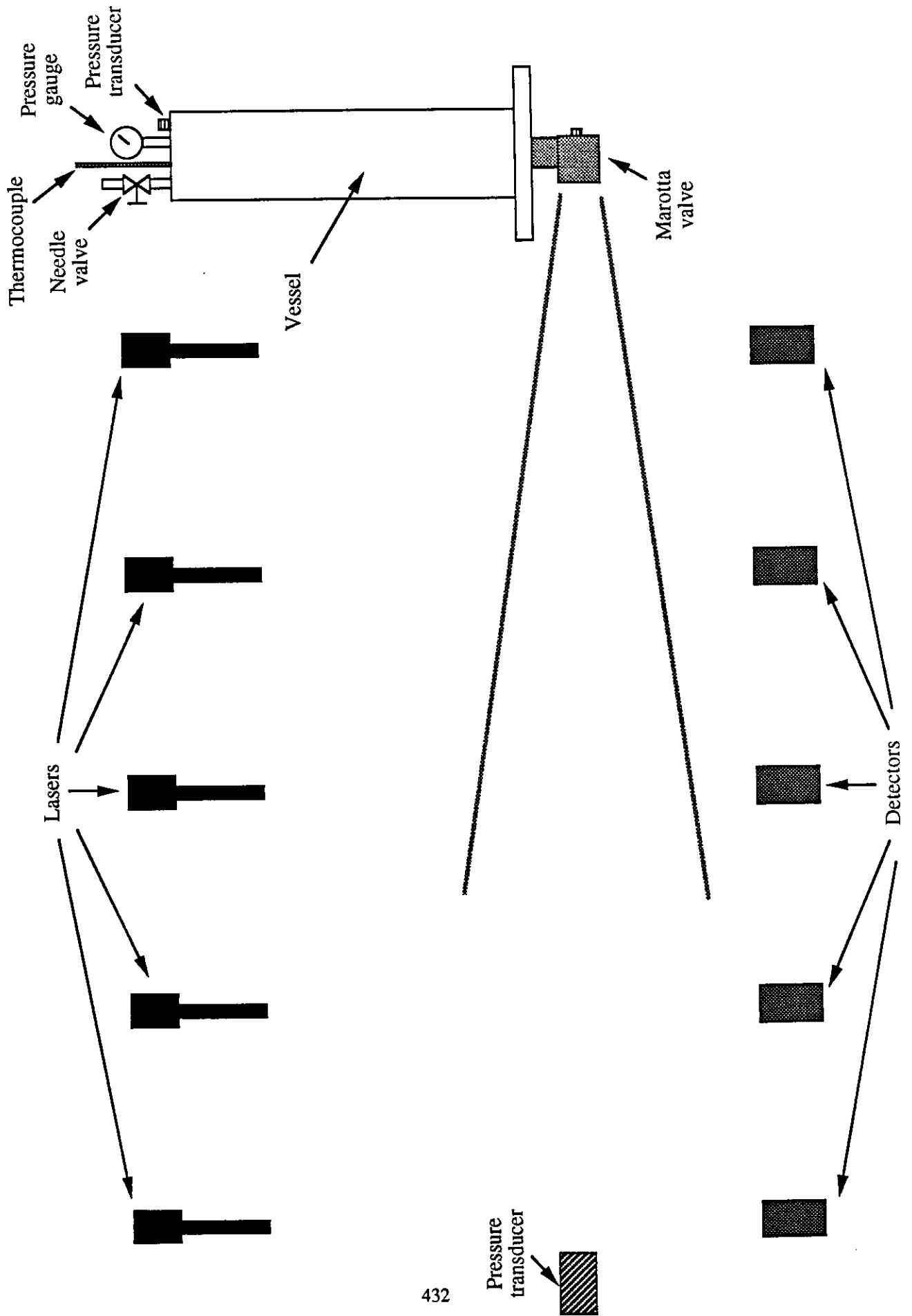
Since the work is still in progress, only the **data** from FC-218 discharges will be presented. The results were obtained using a cylindrical vessel (with sight glasses and equipped with a **Marotta** valve) at room temperature and 2/3 fill (by volume) condition. Nitrogen charged pressures varied from about 2.75 to 4.12 MPa.

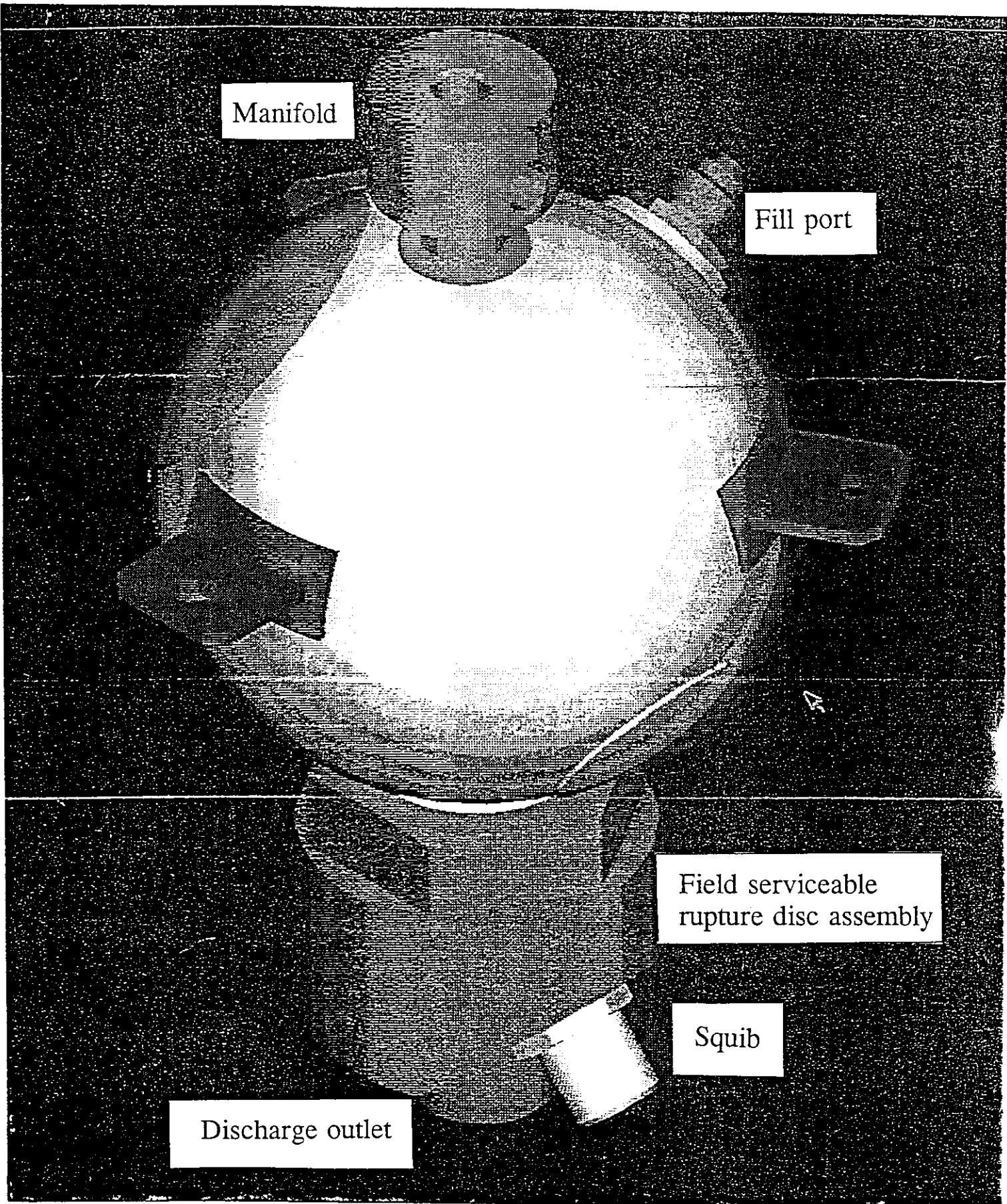
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# EXPERIMENTAL APPARATUS





Manifold

Fill port

Discharge outlet

Field serviceable  
rupture disc assembly

Squib

Spherical Vessel

# Experimental Conditions

## Agents

- HFC-125, FC-218, & CF<sub>3</sub>I

## Initial conditions

- Different fill densities
- Different nitrogen charge pressures

## Test conditions

- High & low temperatures
- Different release mechanisms
- Different bottle shapes (cylindrical vs. spherical)

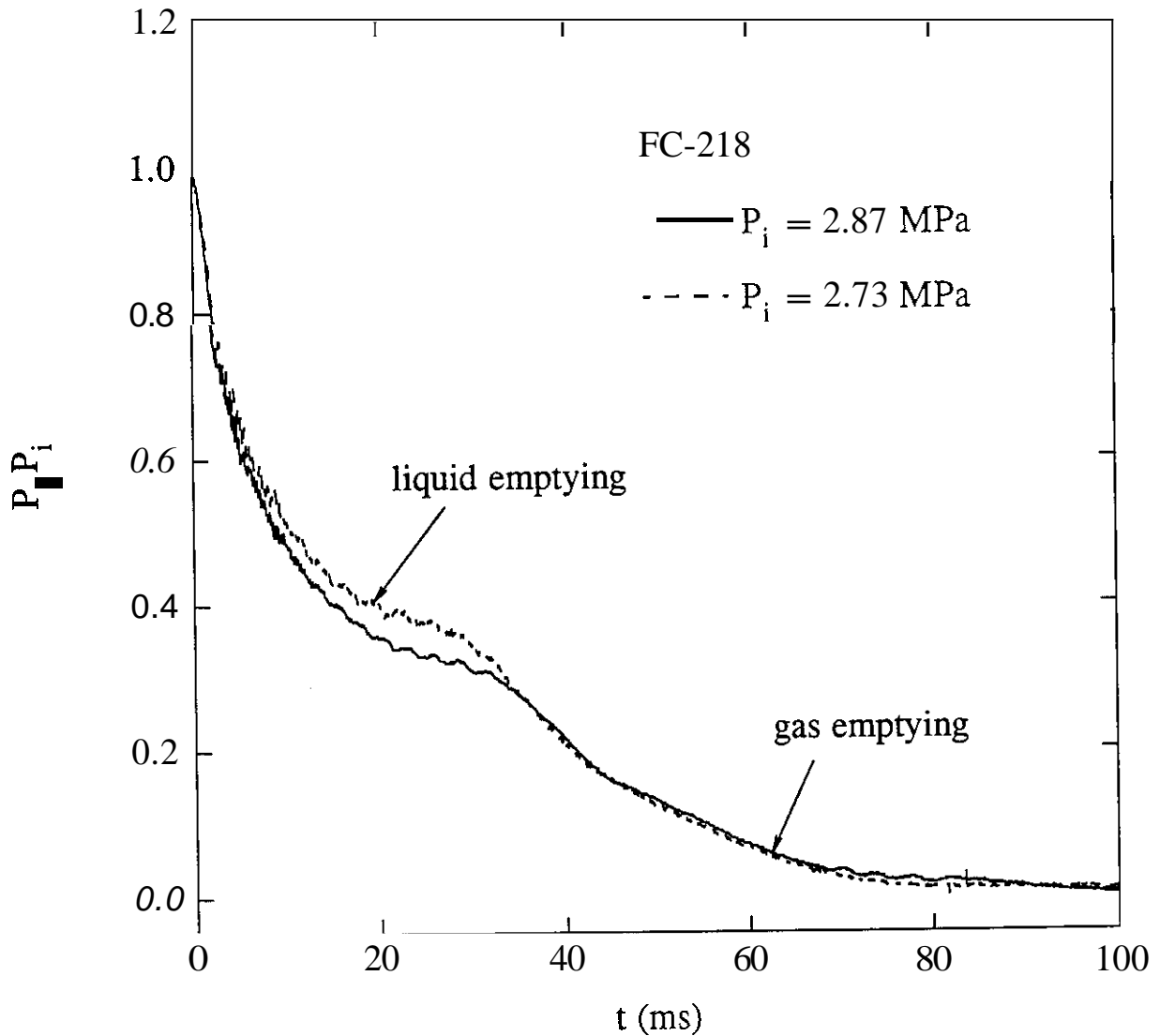
## Parameters to be measured

- Temporal variations of internal pressure and temperature
- Documentation of discharge by high speed photography
- Documentation of depressurization processes inside the vessel (using a vessel with sight glasses)
- Average velocities using laser attenuation technique

$$average\ velocity = \frac{(X)_{laser\ a} - (X)_{laser\ b}}{t_{extinction,\ a} - t_{extinction,\ b}}$$

- Downstream dynamic pressures

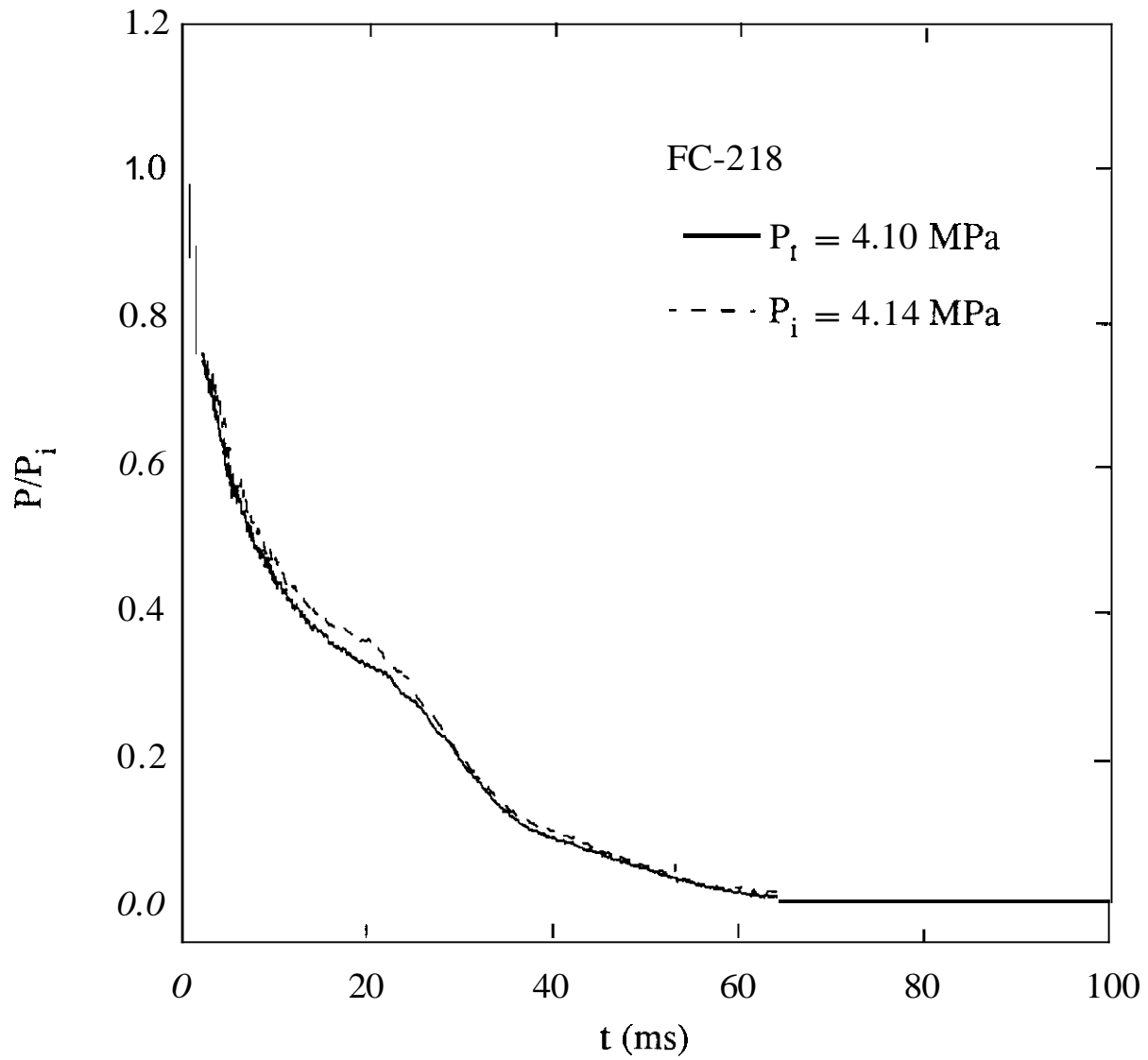
Temporal variations of pressure inside the vessel during discharge at room temperature

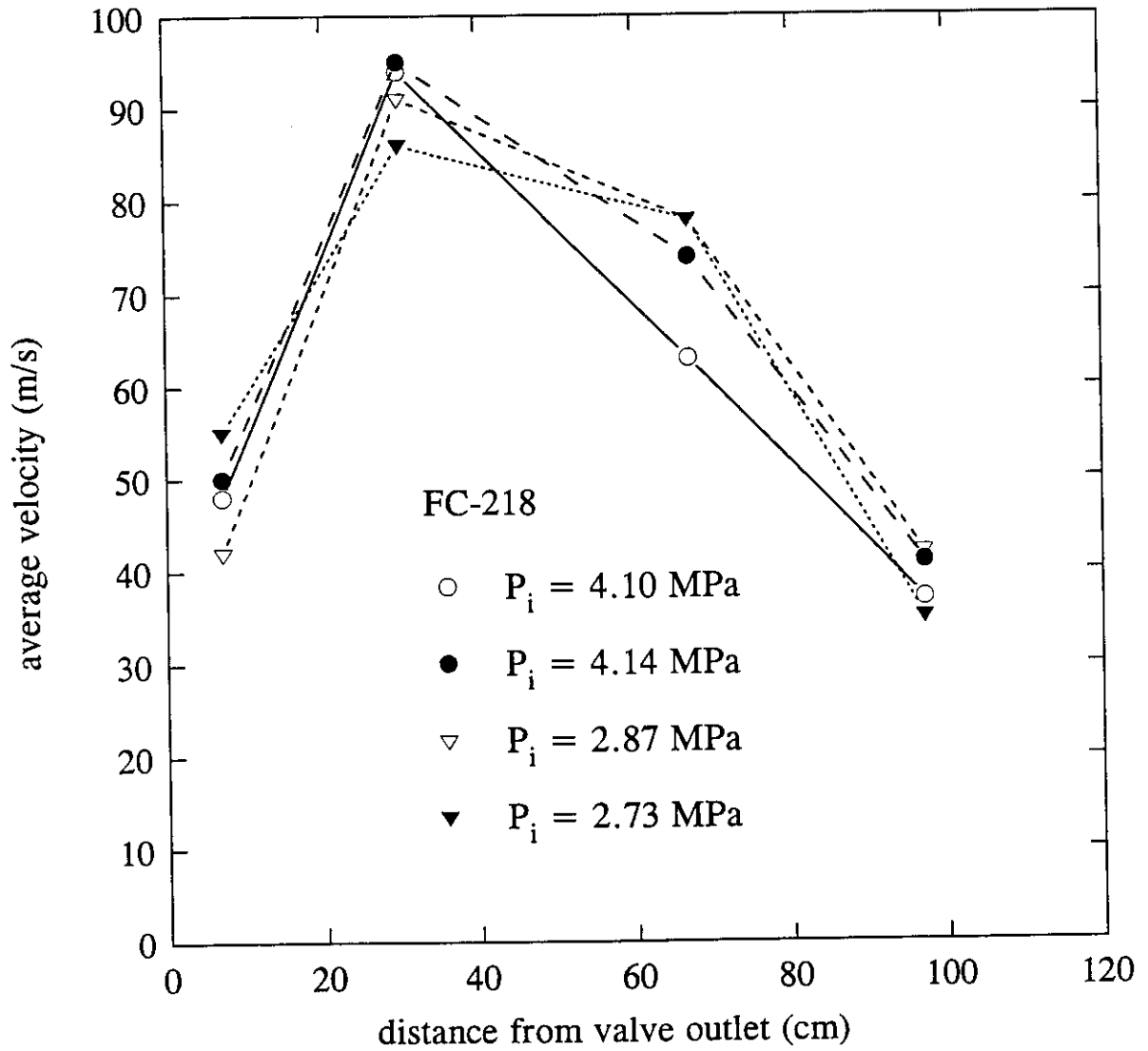


*Note that degassing of nitrogen was not observed inside the vessel. In our case,*

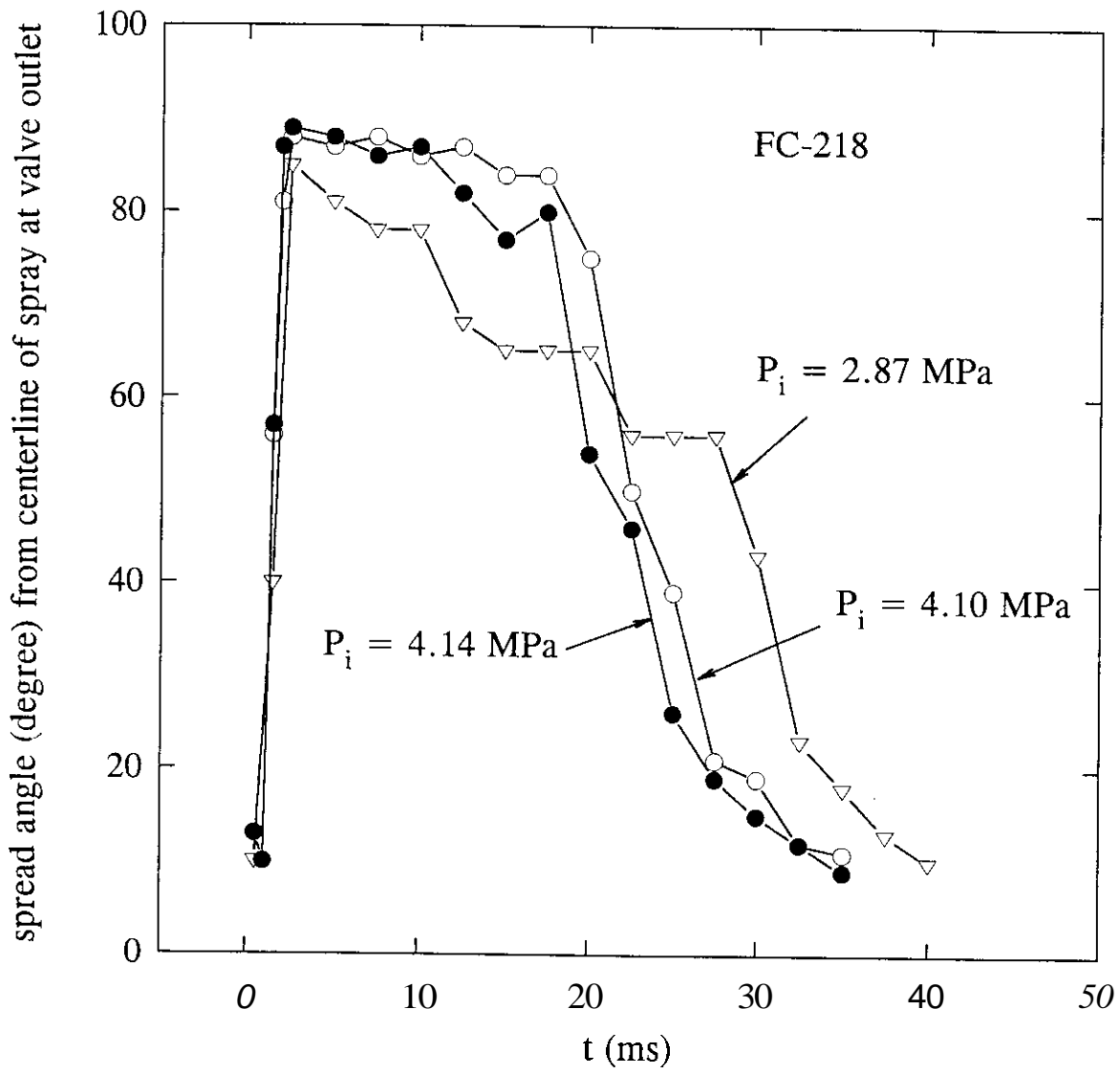
$$t_{discharge} \ll t_{degas}$$

Temporal variations of pressure inside the vessel during discharge at room temperature









Information from the downstream pressure transducer can be used to obtain some qualitative two-phase behavior of the flashing spray

If one assumes the two-phase flow to be homogeneous, then

$$\mathbf{M} \text{ (momentum flux)} = \frac{\mathbf{G}^2}{\rho_h} = \rho_h u^2$$

with

$$\rho_h = \rho_g \alpha + (1 - \alpha) \rho_l$$

where  $G$  is the mass flux,  $u$  is the average velocity,  $\alpha$  is the void fraction,  $\rho_h$  is the homogeneous mixture density, and  $\rho_g$  and  $\rho_l$  are the saturated vapor and liquid densities

Solve for  $\alpha$ , then

$$\alpha = \frac{\rho_l}{\rho_l - \rho_g} \left( 1 - \frac{\mathbf{M}}{\rho_l u^2} \right)$$

If  $M$  (measured from the downstream transducer) and  $u$  (measured from laser extinction) are known, one could obtain an estimate of  $\alpha$  by assuming the spray is at the boiling point of the agent.